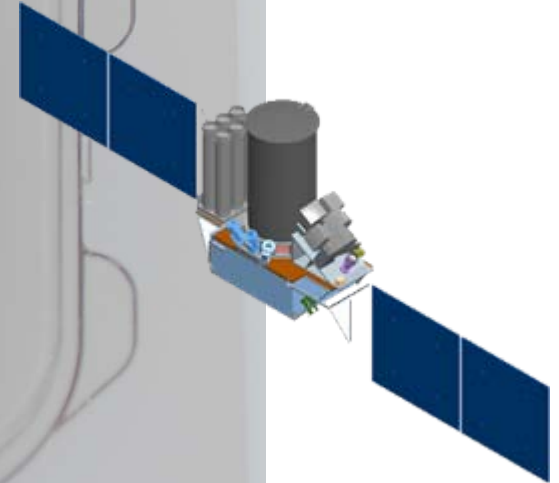


# eROSITA

Norbert Meidinger

## Outline

- I. eROSITA project
- II. Telescope
- III. pnCCD camera
- IV. Characteristics
- V. Outlook



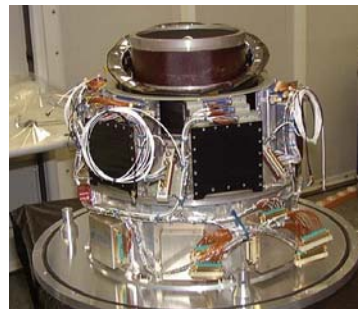
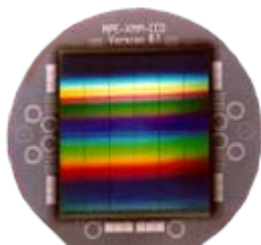
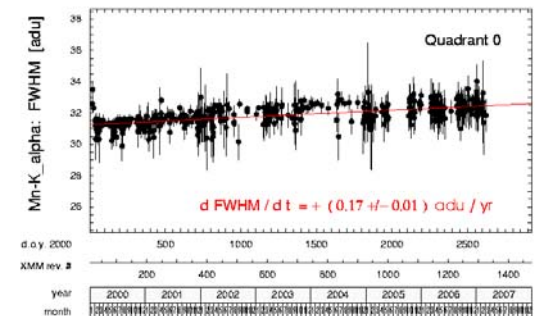
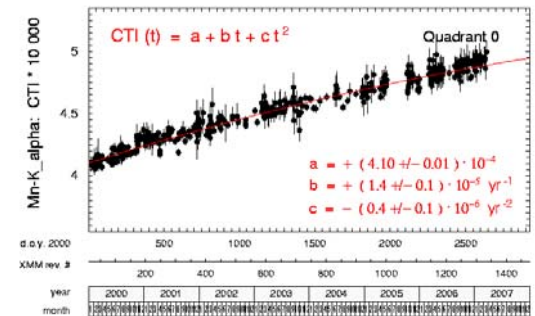
# I. eROSITA background



pnCCD development for **spectroscopy and imaging of X-rays**:  
 → ESA's **XMM-Newton** X-ray astronomy satellite (12/1999):  
**XMM-pnCCD** X-ray camera (by MPE & HLL)

- X-ray detection → E = [0.3 keV; 10 keV] with QE ≥ 90%
- Image area → 6 cm x 6 cm (full frame mode)
- Pixel size → 150 x 150 μm<sup>2</sup> pixel
- Time resolution → 73 ms (fullframe, standard)
- Noise → 5 electrons rms @ -90°C
- Energy resolution → FWHM(5.9keV)=155eV (single events!)
- Long-term stability → after >7 years in space:  
FWHM(5.9 keV) = 155 eV → 161 eV

XMM-Newton operation extended until 2008, probably 2010



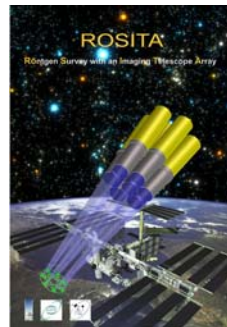
# I. eROSITA project: Background

**XMM-Newton**  
1999 –  
Pointed observations



XMM pnCCD  
camera

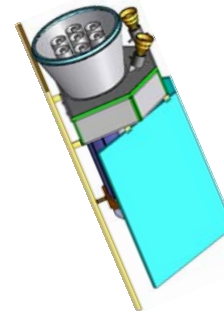
**ROSITA**  
2002



ABRIXAS science  
on the  
International  
Space Station

7 cameras  
256 x 256 pixel  
FS pnCCD

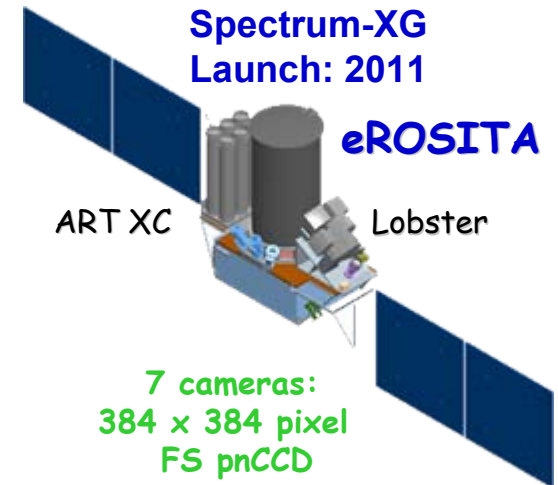
**DUO**  
2004



Dark Energy  
10<sup>4</sup> Clusters  
of Galaxies

7 cameras:  
256 x 256 pixel  
FS pnCCD

**Spectrum-XG**  
Launch: 2011



**eROSITA**

7 cameras:  
384 x 384 pixel  
FS pnCCD

ABRIXAS & DUO science  
10<sup>5</sup> Clusters of Galaxies

DLR grant 04/2007

**ROSAT**  
1990-1998



First X-ray  
all-sky survey  
with an  
imaging telescope

PSPC  
gas counter  
camera

**ABRIXAS**  
1999



To extend the  
all-sky survey  
towards  
higher energies

XMM pnCCD  
camera  
† satellite  
power system

NASA SMEX  
proposal  
† 01/2005

# I. eROSITA project



**eROSITA** = ROSITA + DUO

(extended ROentgen Survey with an Imaging Telescope Array)

Main scientific goals:

First imaging **all-sky survey** (1 y exposure time) 0.3 keV **up to 10 keV**

→ systematic detection of all obscured accreting Black Holes in nearby galaxies and new active galactic nuclei in the hard band.

**Wide and deep survey** (3 y + 0.5 y): observation of dedicated sky regions to detect **100.000 clusters of galaxies**.

Follow-up **pointed observations** (0.5 y) to map out LSS in Universe

→ nature of **Dark Energy and Dark Matter** (96% of Universe) ; model of inflation + re-observation of interesting targets

⇒ total: 5 y

**Satellite: Spectrum-XG**

Launch: 2011, Baikonur

600 km circular orbit, 30° inclination, 96 min. orbit

Launch vehicle: Soyus-2/Fregat

## II. eROSITA project



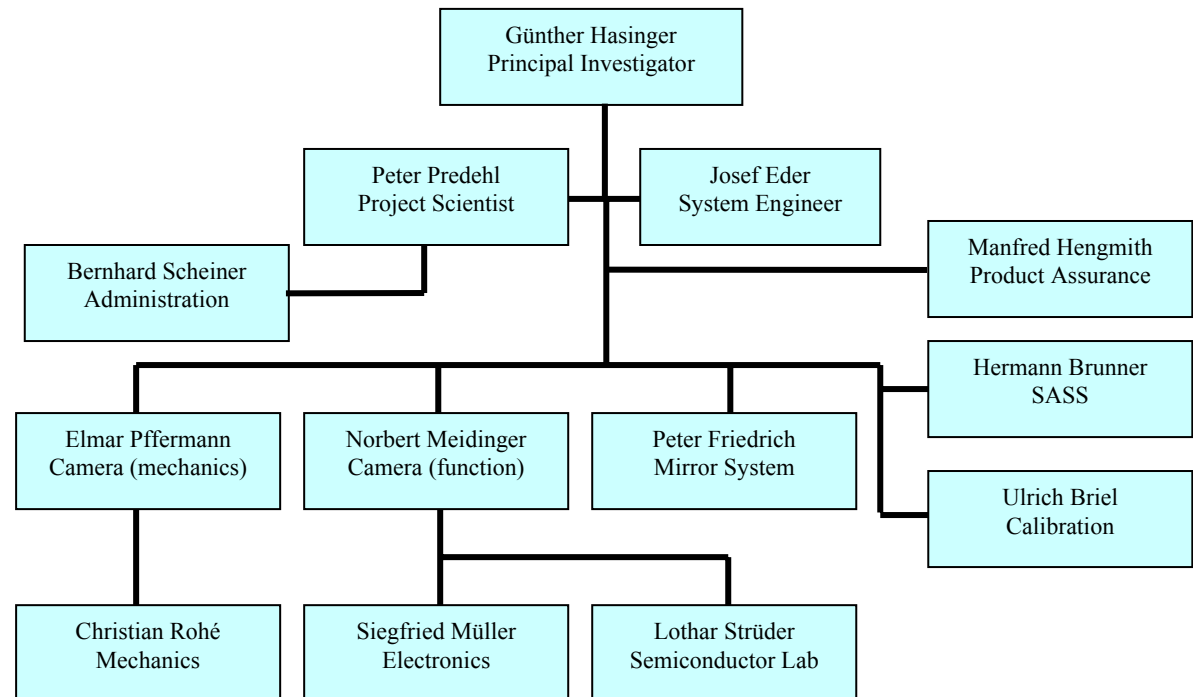
04/2007:

- **Memorandum of understanding** signed by **DLR** and **ROSKOSMOS** for cooperation: eROSITA aboard Russian satellite Spectrum-X-ray-Gamma
- **Funding by DLR granted** (until launch): **21 Mio. €**

### Consortium:

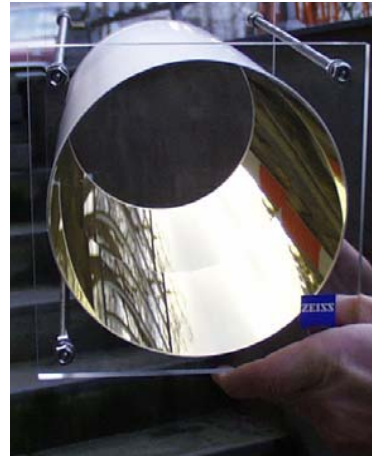
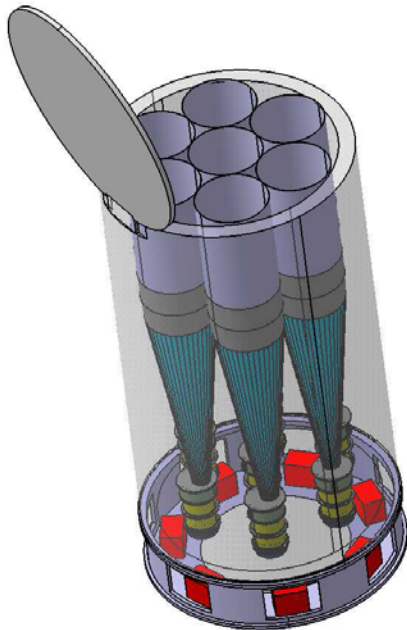
led by: **MPE**

IAAT (Tübingen),  
FAU (Erlangen),  
AIP (Potsdam),  
UH (Hamburg),  
MPA (Garching),  
IKI (Moscow)



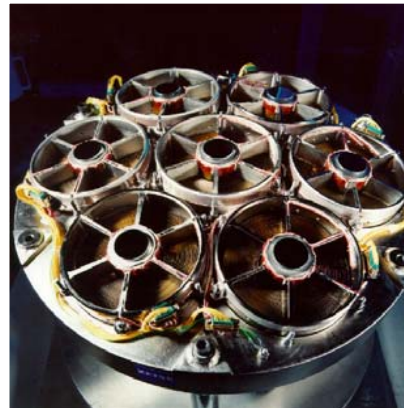
## II. Telescope

Telescope =  
7 × (mirror system  
+ pnCCD camera in focus)



mirror shell

**ABRIXAS:**

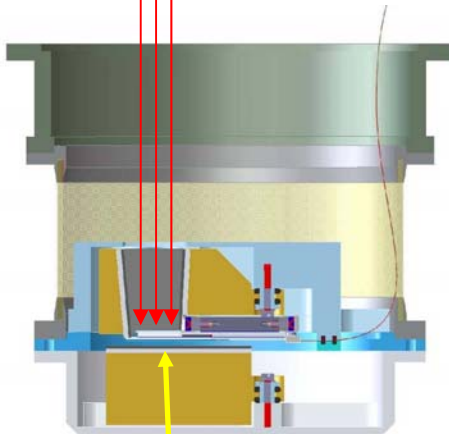


7 mirror systems

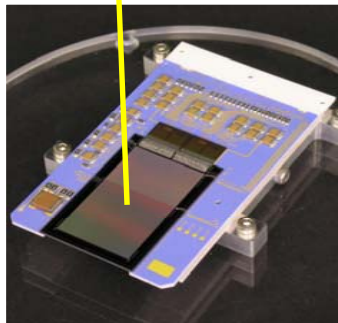
- 7 mirror systems with parallel optical axes
- 54 nested mirror shells Wolter-I type (ABRIXAS 27 shells qualified)
- focal length: 1.6 m
- FoV:  $\varnothing = 1^\circ$
- angular res.  $< 15''$  HEW (1 keV, on-axis)
- gold coating on nickel shell
- nickel-galvanoplasting process (similar XMM-Newton)
  - Media Lario Srl
  - + Carl Zeiss AG
  - + Kayser-Threde GmbH

## II. Telescope

7 identical pnCCD cameras  
active cooling → TECs + radiator  
graded shield → background



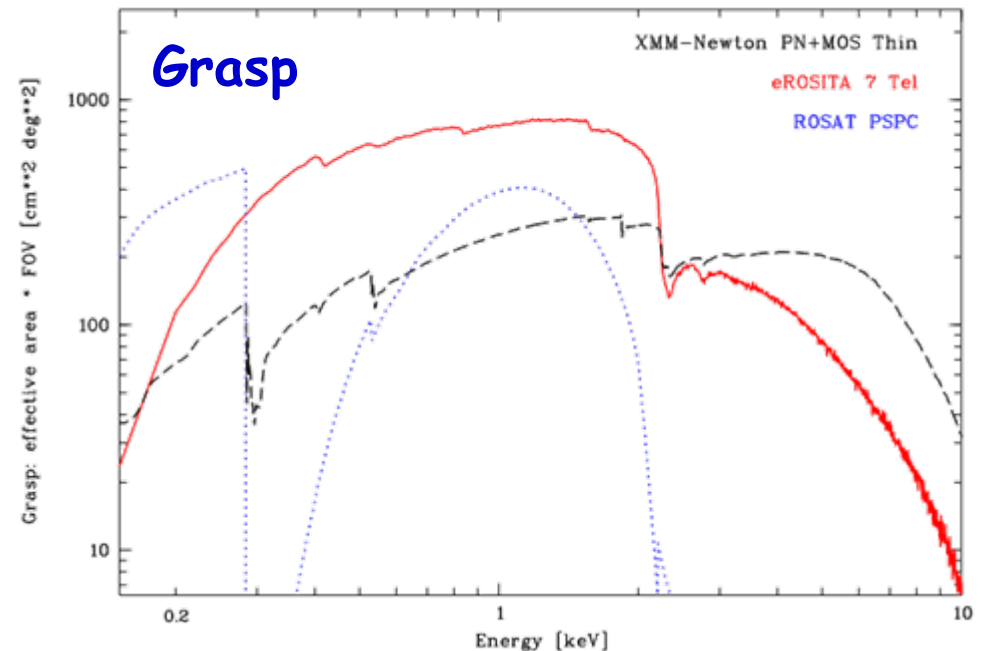
pnCCD camera



pnCCD detector module

### Detector requirements:

- ◆ Image area 28.8 mm × 28.8 mm  
= 384 × 384 pixel → FoV:  $\varnothing = 1^\circ$
- ◆ Pixel size: 75  $\mu\text{m}$  × 75  $\mu\text{m}$
- ◆ Time resolution: 50 ms ↔ frame rate: 20 images / s
- ◆ Energy range: 0.3 keV - 10 keV
- ◆ Energy resolution  $\approx 138$  eV @ 6 keV
- ◆ Power consumption in f.p.  $\approx 0.6$  W
- ◆ Operating temperature  $\approx -80^\circ\text{C}$
- ◆ On-chip optical + UV filter
- ◆ Long term stability > 5 y
- ◆ ...



### III. pnCCD camera

#### pnCCD:

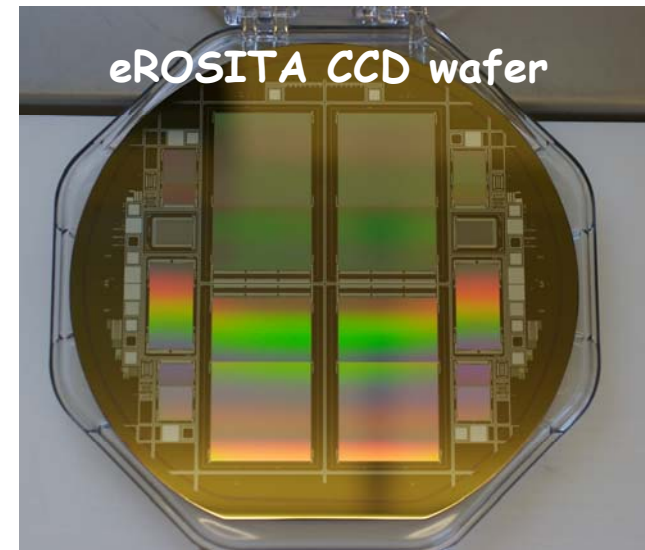
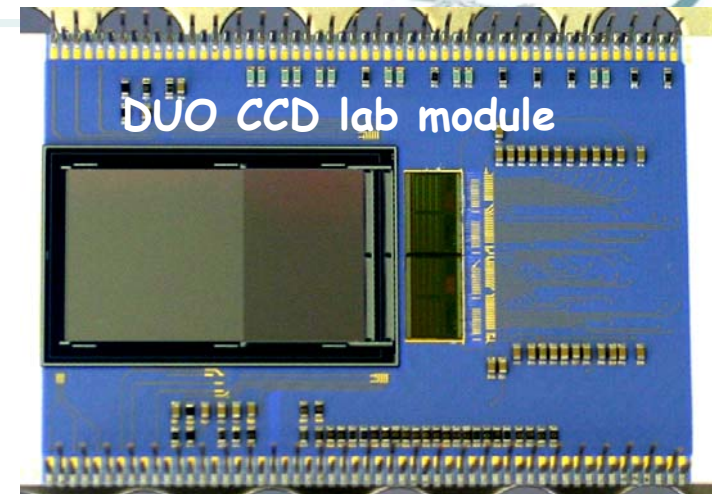
DUO pnCCDs (C11) = prototype CCDs of eROSITA  
256 x 256 pixel, BCB, light filter  
tested extensively 2005 - 2007

eROSITA flight pnCCDs (C12):  
production started 07/2006 - 10/2007 (?)  
standard process technology - low risk  
changes:

- number of pixels
- MOS gate to control charge reset

9 wafers x 4 pnCCDs

- 3 lab/breadboard + 2 QM + 7 FM  
+ 4 FS CCD modules





### III. pnCCD camera

#### pnCCD:

selection by **Kaltprober**:

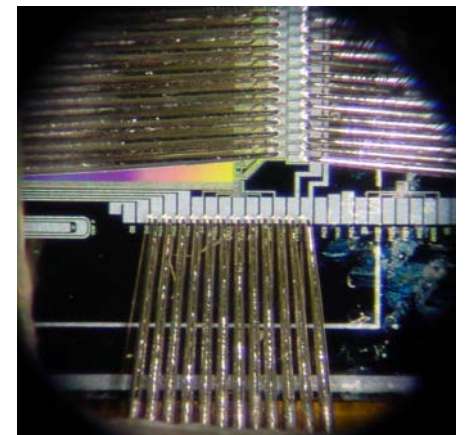
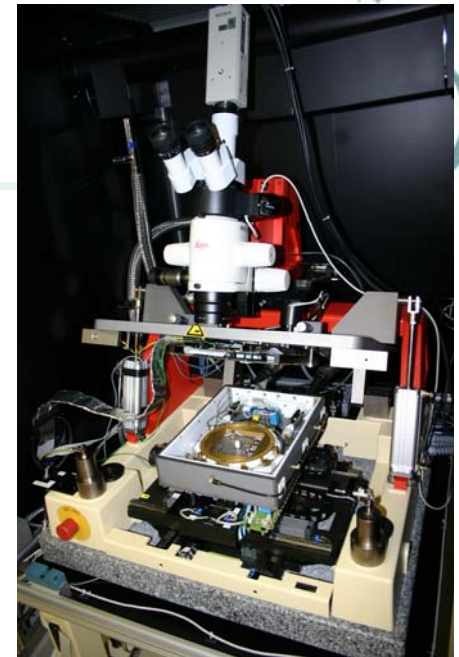
- spectroscopic performance tests  
just with wafer/chip - no module
- (Necessary: Probecard + single needles → alignment,  
darkness, cooling, Fe55 source, complete electronics, DAQ system)
- **successful** test with a DUO CCD wafer
- **modification** for larger eROSITA CCDs

**CAMEX** ASICs: readout of eROSITA pnCCD  
fabrication started 02/2007 - 06/2007 (?)

changes:

- 6 → 8 `` wafer, new process techn., e.g. thinner gate oxides
- linearity + speed of output buffer + bandwidth
- control of ref. currents by CAMEX programming
- 2 CAMEX types: conservative + innovative
- fallback: present CAMEX chips

→ 25. April: CAMEX Electronics  
(Sven Herrmann)



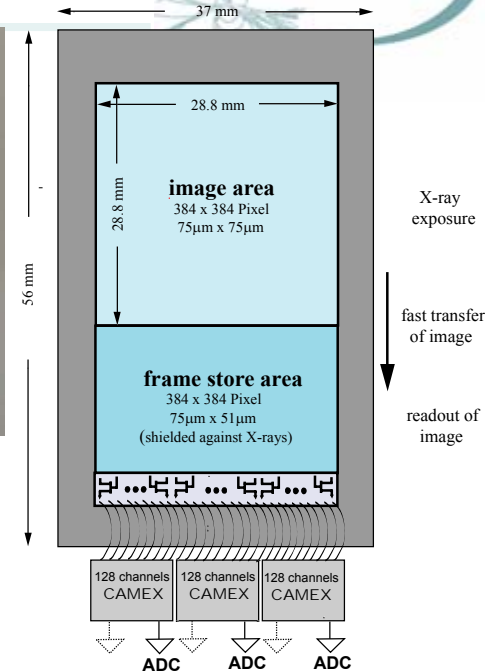
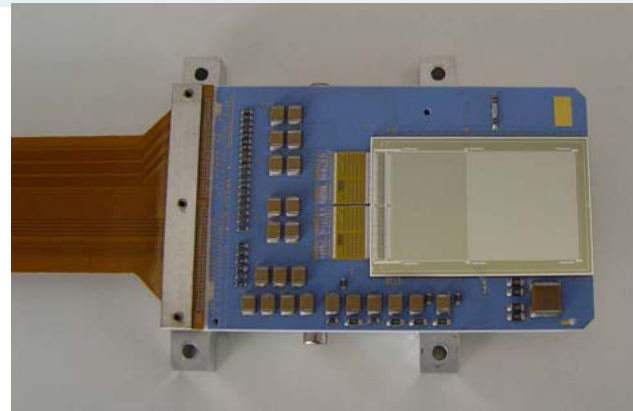
### III. pnCCD camera

#### pnCCD module:

##### Ceramic PCB:

5+1 layers; thick film technology;  
CCD, CAMEXes, SMD C's + R's  
wedge bonding

used + successfully tested for 20 y  
esp. in all lab. pnCCD detector tests  
→ adopted for **eROSITA flight camera**

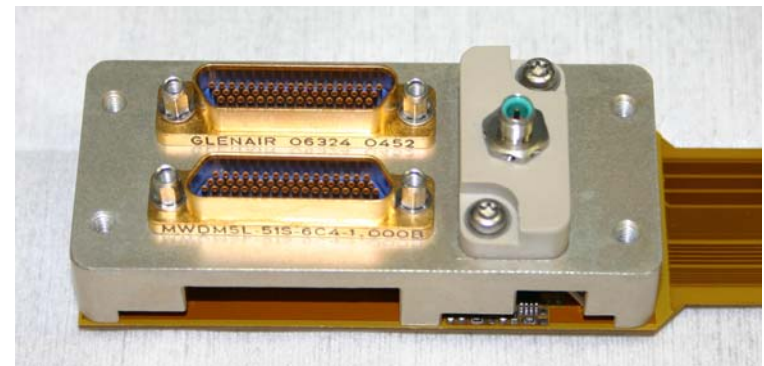


#### Flex lead:

- connects CCD module to camera electronics  
→ wedge bonds and soldering to connectors
- analog CCD clock pulses generated on flex
- analog output buffer (ADC)

#### Status prototype pnCCD module

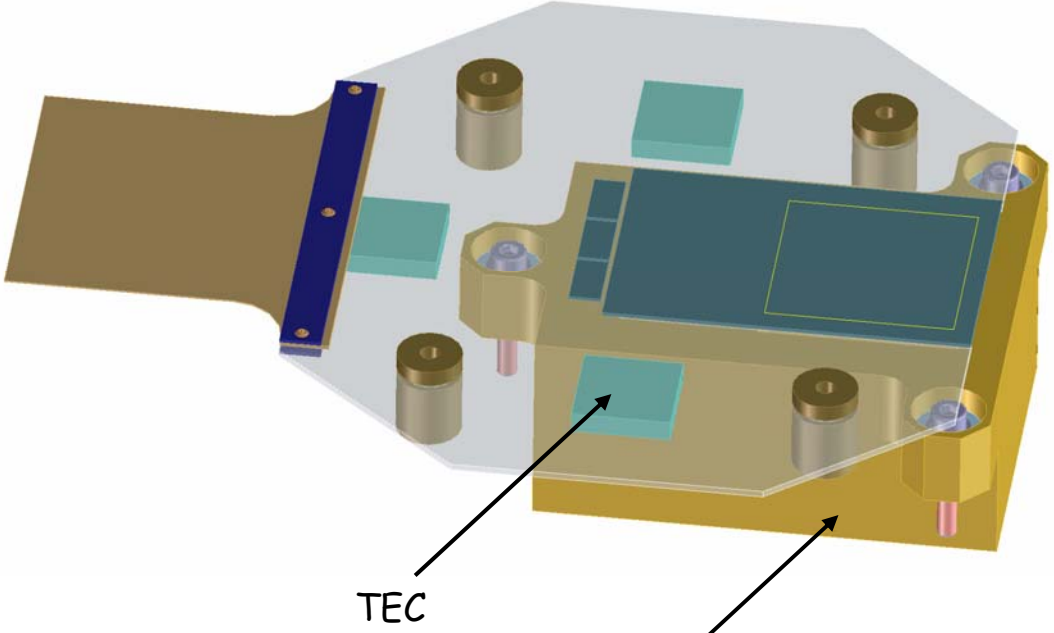
- tested with CAMEX
- soon with CCD



# III. pnCCD camera

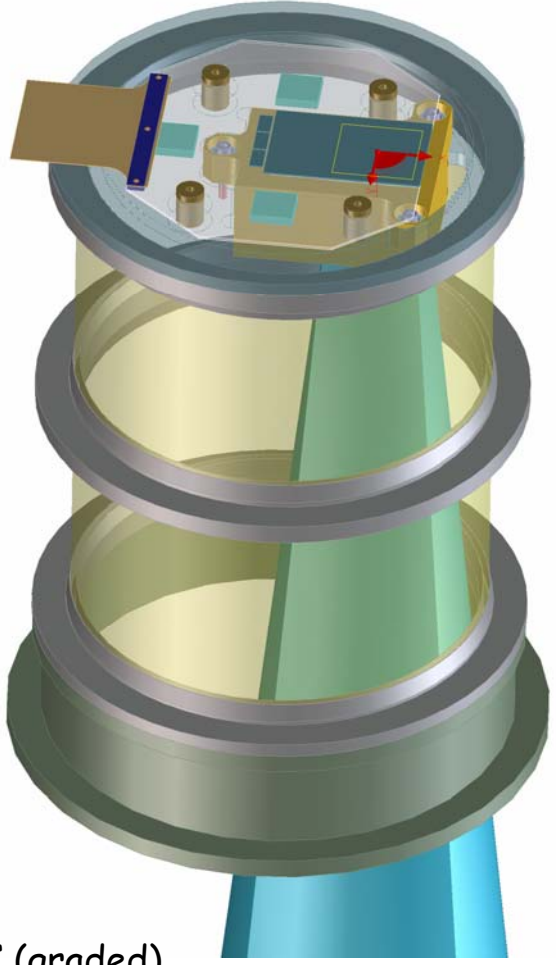


pnCCD camera:  
pnCCD module + mechanical + thermal



TEC

shielding: Cu (proton) + Al<sub>2</sub>O<sub>3</sub> + BC (graded)

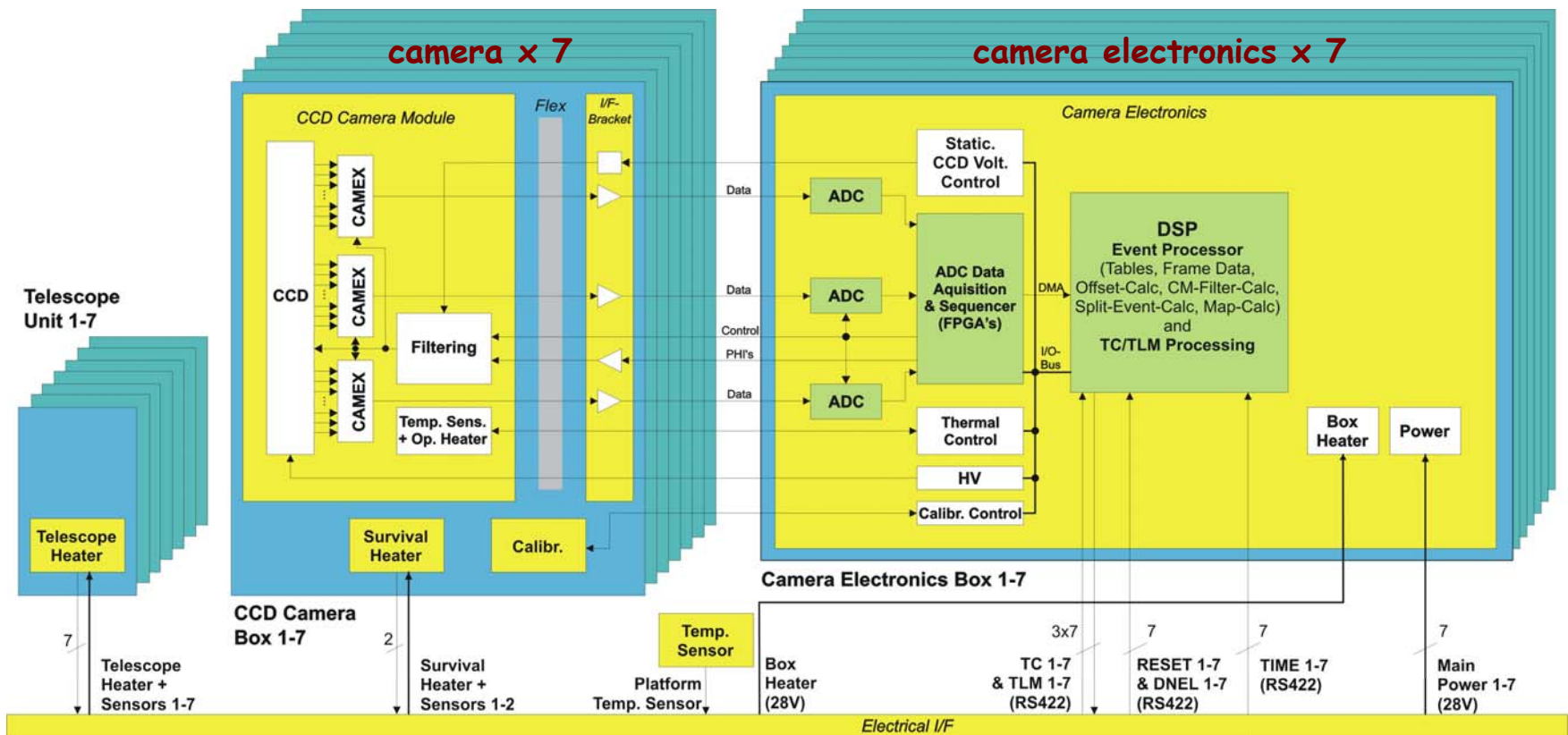


# III. pnCCD camera



## 7 x camera electronics (outside camera)

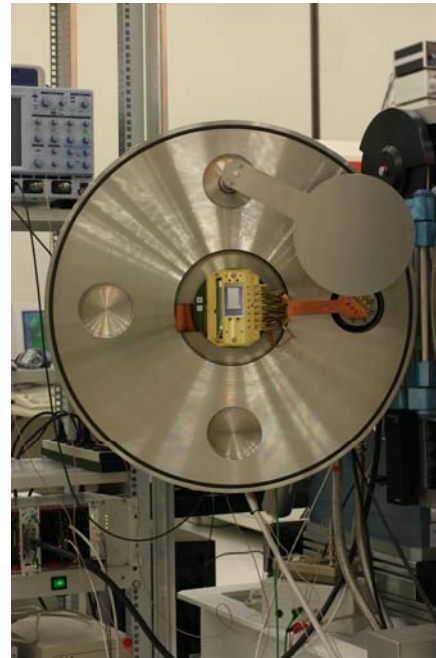
ADCs, DSP event analyzer, power supplies, FPGA sequencer, HK controller, ...



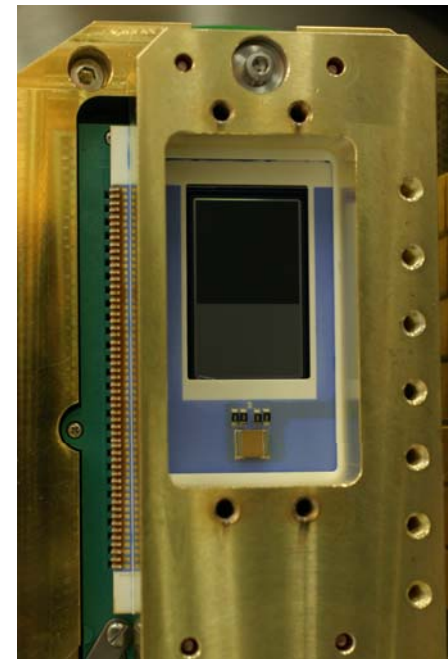
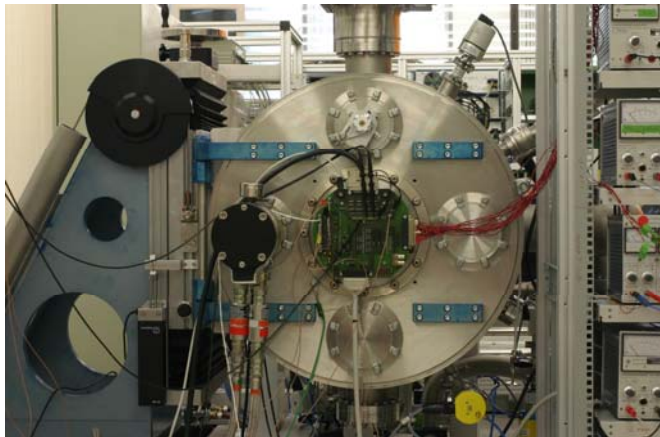
eROSITA Block Diagram 384\*384 CCD (V1.0a, 12.12.2006/Siegfried Müller)

→ lab DAQ → TRoPIC camera at PANTER for telescope cal. → flight prototype → flight

## IV. Performance

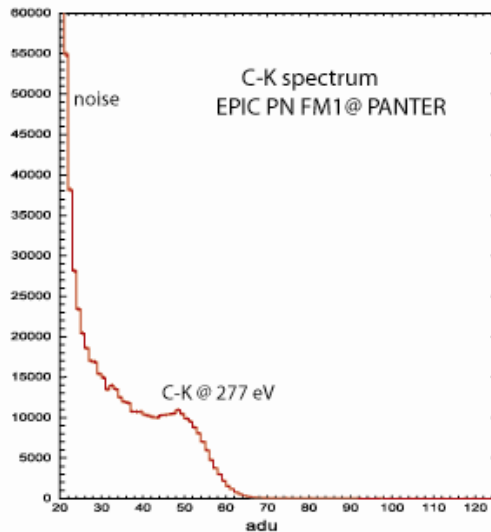


**ROESTI @ HLL:**  
performance tests of  
lab. prototype pnCCD modules  
256 x 256 pixel CCD



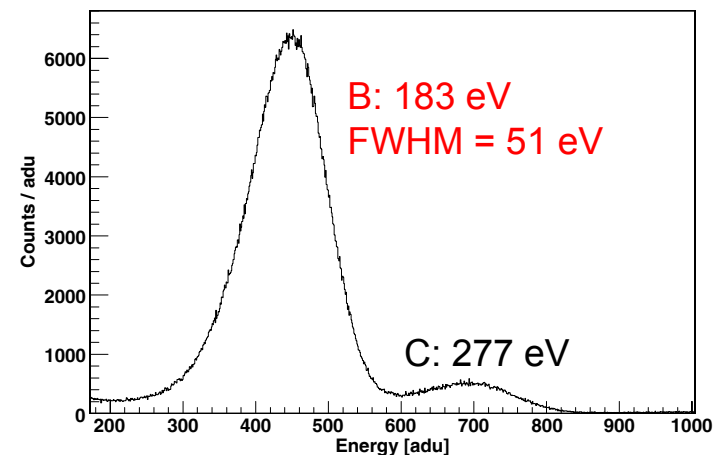
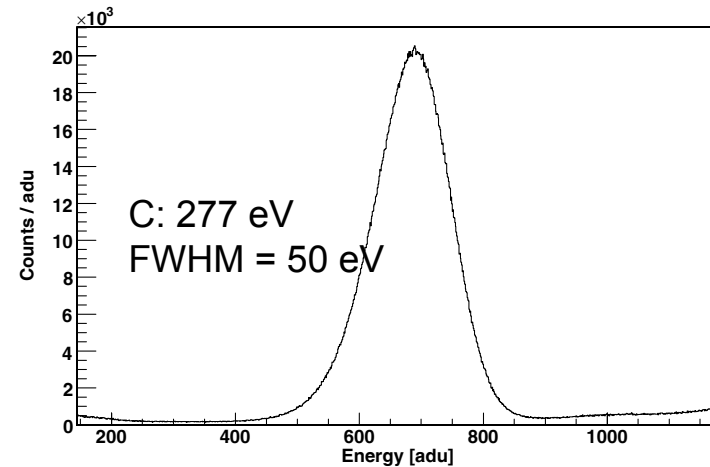
## IV. Performance

### XMM-Newton: pnCCD



24. April:  
lower energy - same pnCCD →  
„pnCCDs at FLASH“ (Christian R.)  
„Data analysis“ (Robert A.)

### eROSITA: frame store pnCCD



„eROSITA conditions“:

T = -80°C; frame rate 20 images/s; light filter

## IV. Characteristics



pnCCD	eROSITA	XMM-Newton
# Cameras	7	1
# pixels per pnCCD	294,912	153,600
Total # pixels	2,064,384	153,600
Operation mode	frame store	fullframe
Out of time (image smearing)	0.4 %	6 %
# pixel transfers	[385;768]	[1;200]
Pixel size	75 x 75 $\mu\text{m}^2$	150 x 150 $\mu\text{m}^2$
Sens. depth	450 $\mu\text{m}$	300 $\mu\text{m}$
Operating temperature	-80°C	-90°C
Read noise	2 el. rms	5 el. rms
CTI	$2 \times 10^{-5}$	$50 \times 10^{-5}$
FWHM@5.9keV	all events: 138 eV	singles: 155 eV
FWHM@277eV or 183 eV	all events: 50 eV	-

## V. Outlook



**eROSITA is now a project !!!**

Project with **highest priority** at MPE-group of HLL and X-ray group of MPE

**Cameras** are built **by MPE & HLL** (no industry - in contrast to mirror systems)

**Camera schedule:**

breadboard construction + concept phase	presently
Final design	→ 12/2007
DM manufacturing and assembly + test	→ 06/2008
EQM manufacturing + assembly + test	→ 12/2008
FM1 manufacturing, assembly, test	→ 22.04.2009
FM2 manufacturing, assembly, test	→ 17.06.2009
FM3 manufacturing, assembly, test	→ 12.08.2009
FM4 manufacturing, assembly, test	→ 07.10.2009
FM5 manufacturing, assembly, test	→ 02.12.2009
FM6 manufacturing, assembly, test	→ 27.01.2010
FM7 manufacturing, assembly, test	→ 24.03.2010
Calibration, integration into satellite, final tests	
Satellite launch:	2011

**thanks to the whole team**